

Integration Science and Technology of Advanced Ceramics for Energy and Environmental Applications

M. Singh

Ohio Aerospace Institute
NASA Glenn Research Center
Cleveland, OH 44135, USA

Abstract

The discovery of new and innovative materials has been known to culminate in major turning points in human history. The transformative impact and functional manifestation of new materials have been demonstrated in every historical era by their integration into new products, systems, assemblies, and devices. In modern times, the integration of new materials into usable products has a special relevance for the technological development and economic competitiveness of industrial societies. Advanced ceramic technologies dramatically impact the energy and environmental landscape due to potential wide scale applications in all aspects of energy production, storage, distribution, conservation, and efficiency. Examples include gas turbine propulsion systems, fuel cells, thermoelectrics, photovoltaics, distribution and transmission systems based on superconductors, nuclear power generation, and waste disposal.

Robust ceramic integration technologies enable hierarchical design and manufacturing of intricate ceramic components starting with geometrically simpler units that are subsequently joined to themselves and/or to metals to create components with progressively higher levels of complexity and functionality. However, for the development of robust and reliable integrated systems with optimum performance under different operating conditions, the detailed understanding of various thermochemical and thermomechanical factors is critical. Different approaches are required for the integration of ceramic-metal and ceramic-ceramic systems across length scales (macro to nano). In this presentation, a few examples of integration of ceramic to metals and ceramic to ceramic systems will be presented. Various challenges and opportunities in design, fabrication, and testing of integrated similar (ceramic-ceramic) and dissimilar (ceramic-metal) material systems will be discussed. Potential opportunities and need for the development of innovative design philosophies, approaches, and integrated system testing under simulated application conditions will also be presented.

National Aeronautics and Space Administration



Integration Science and Technology of Advanced Ceramics for Energy and Environmental Applications

M. Singh
Ohio Aerospace Institute
NASA Glenn Research Center
Cleveland, OH 44135

www.nasa.gov

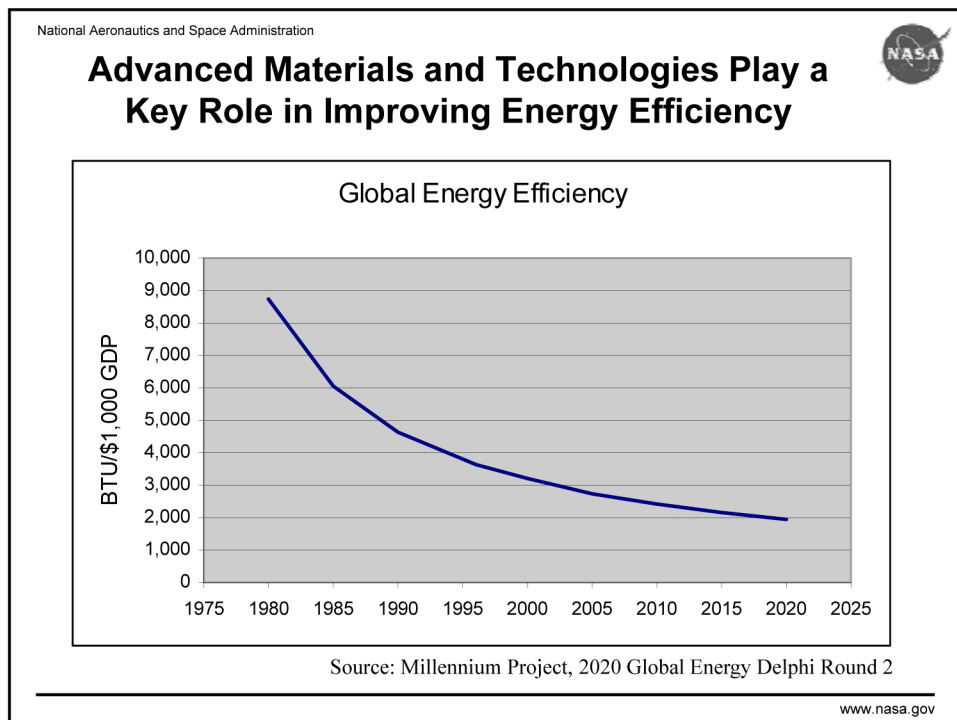
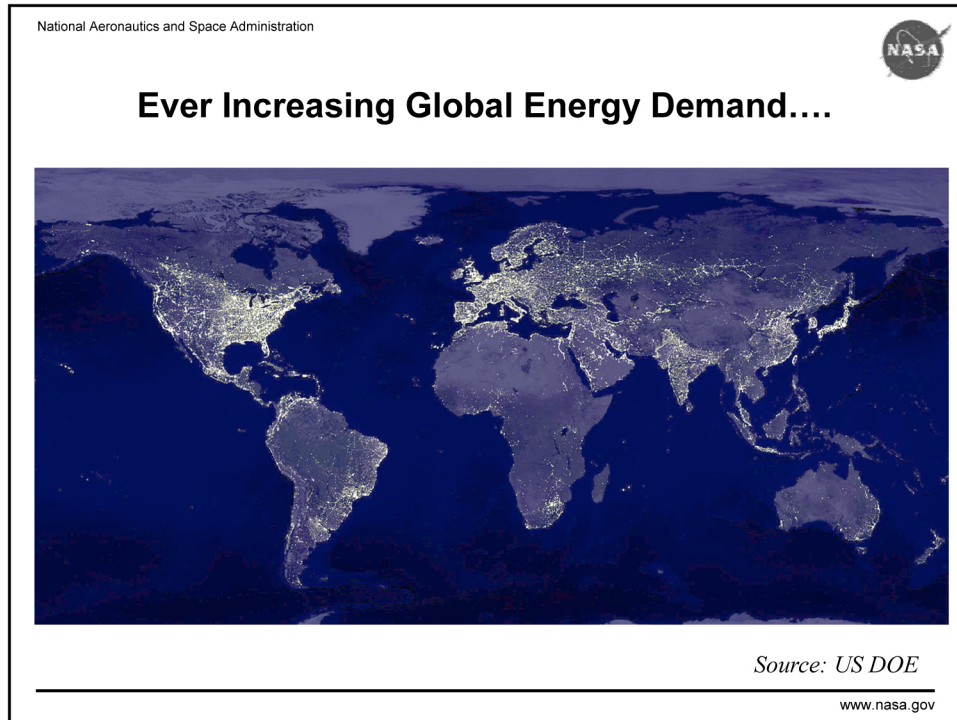
National Aeronautics and Space Administration

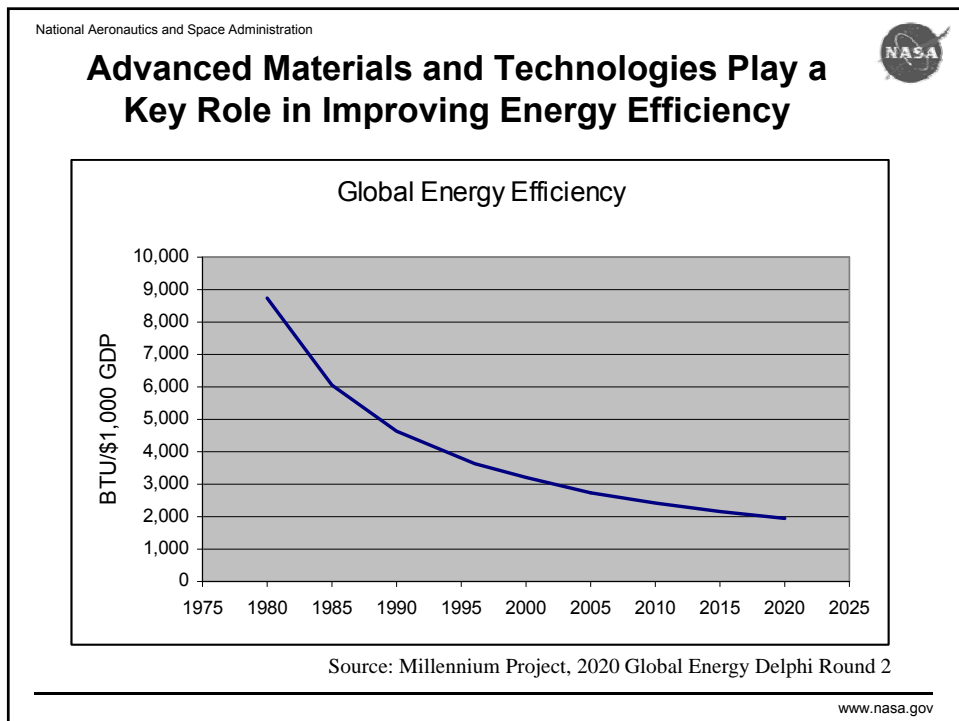
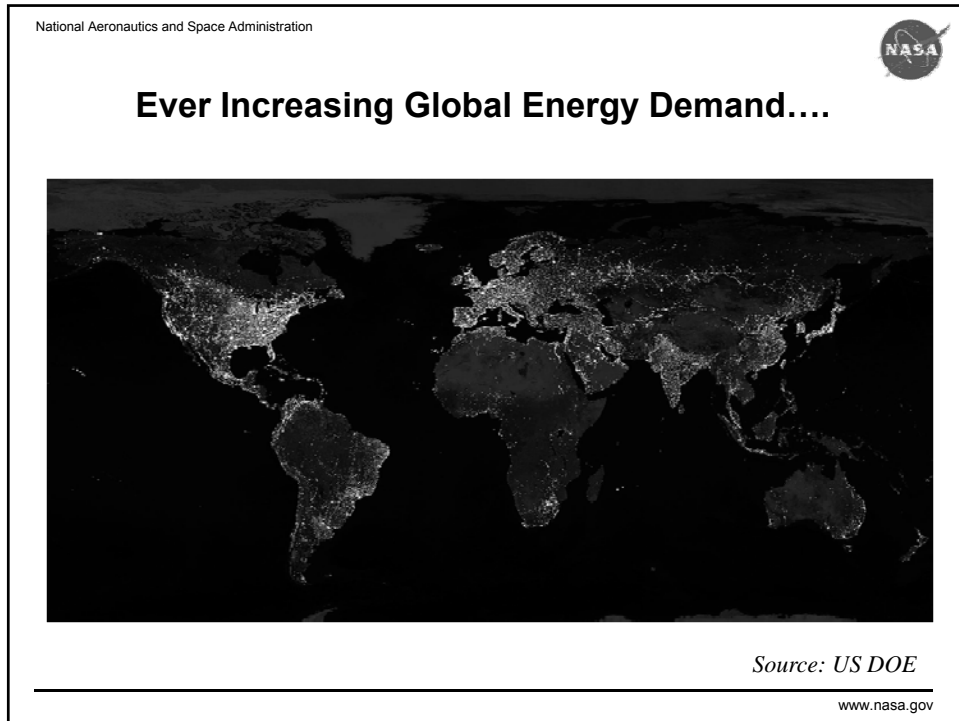


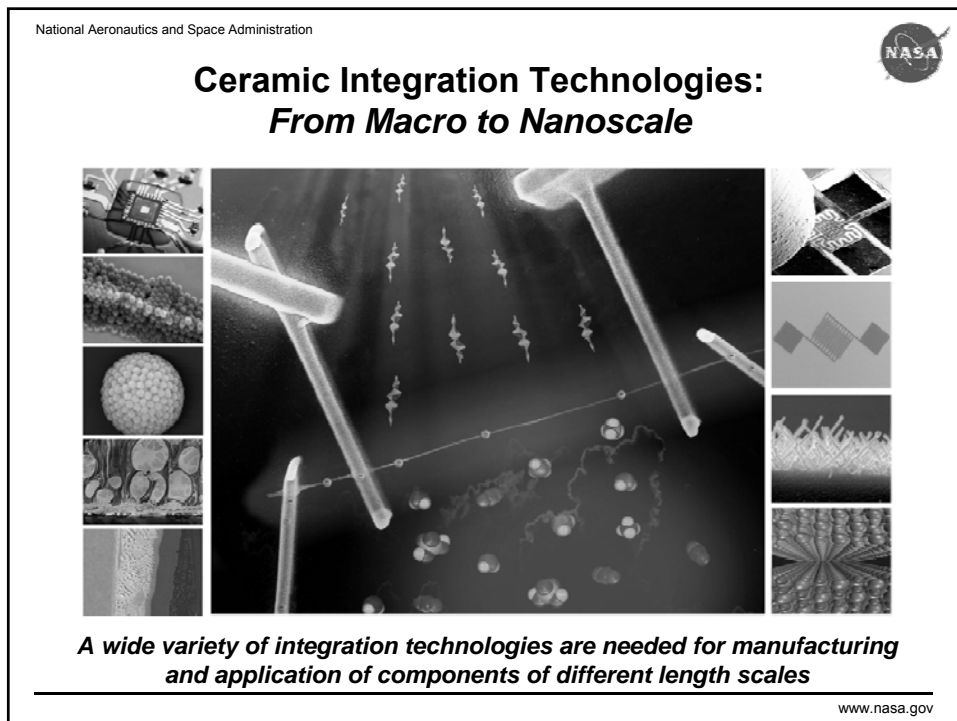
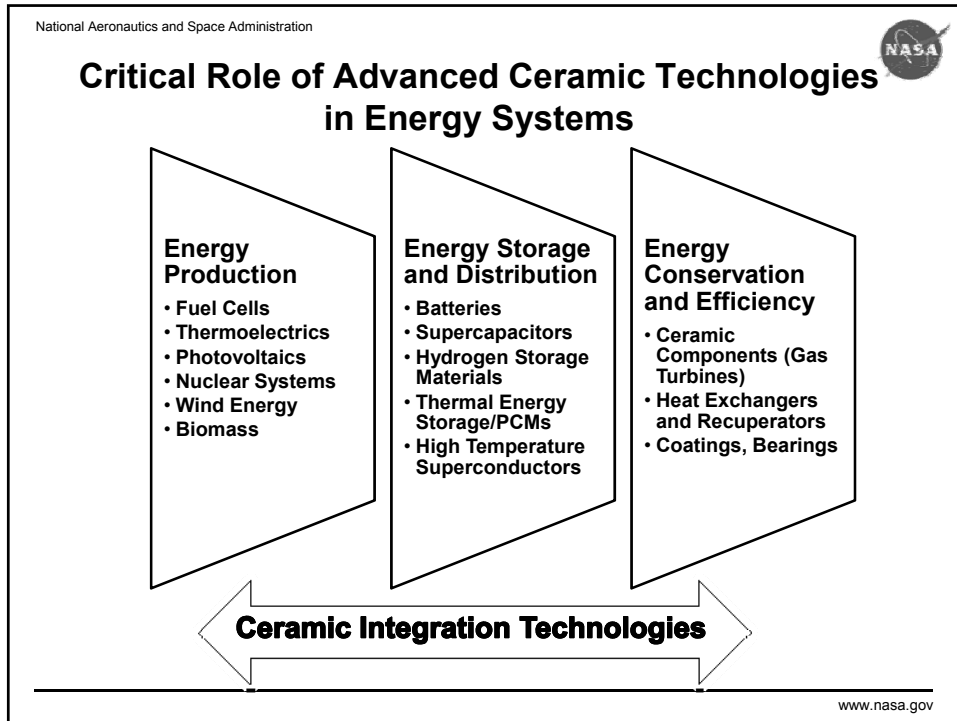
Outline

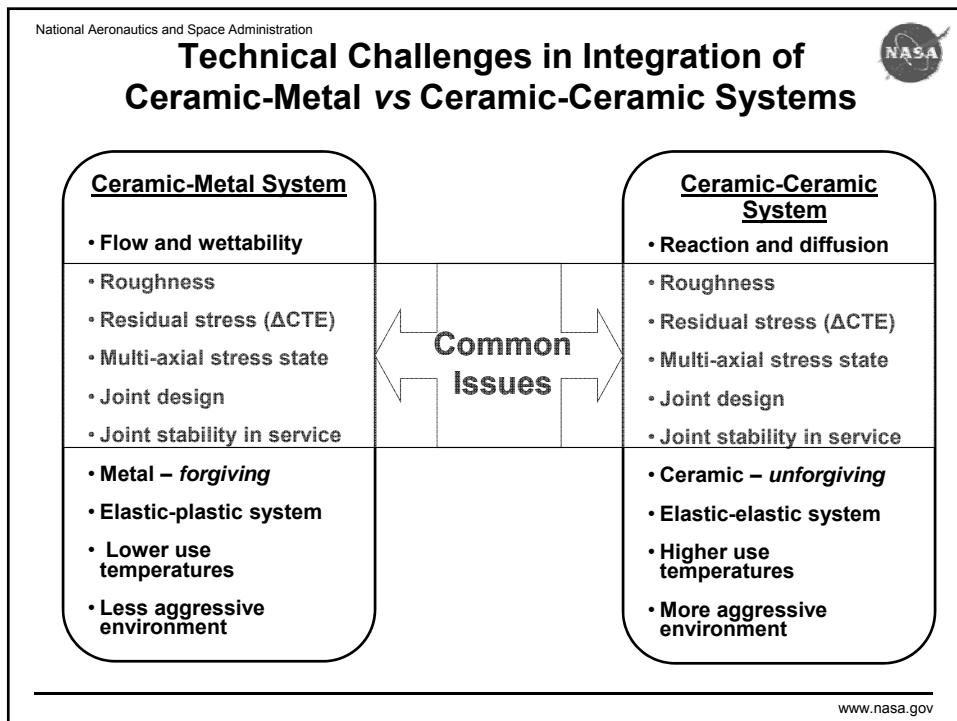
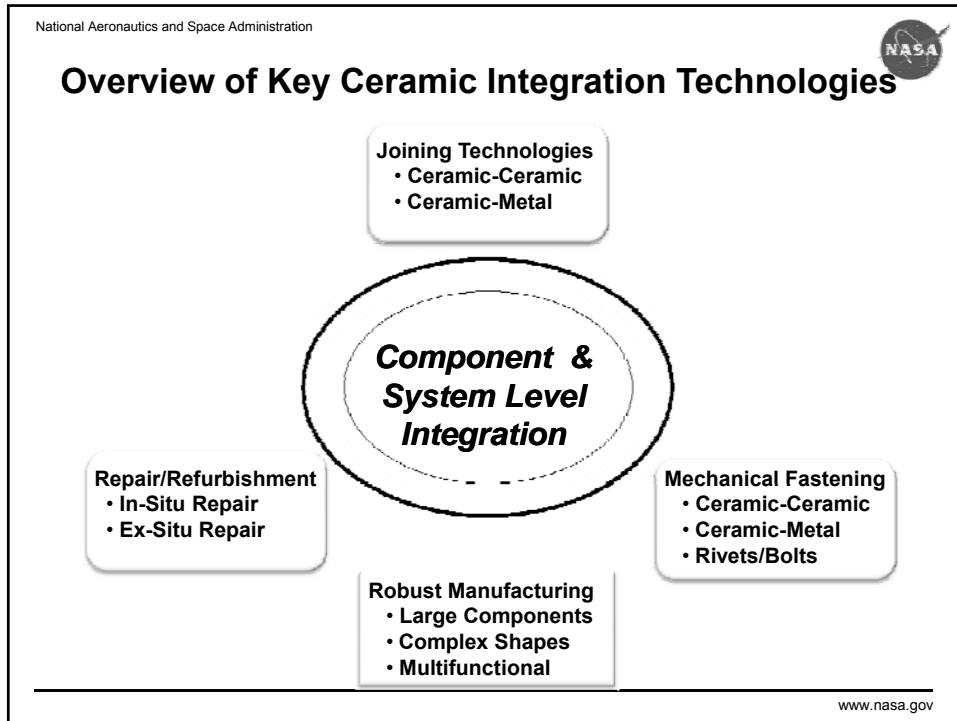
- **Introduction and Background**
- **Technical Challenges in Integration**
 - *Similar vs Dissimilar Systems*
- **Ceramic Integration Technologies**
 - *Wetting and Interfacial Effects*
 - *Ceramic-Metal Systems*
 - *Ceramic-Ceramic Systems*
 - *Testing and Characterization*
- **Concluding Remarks**

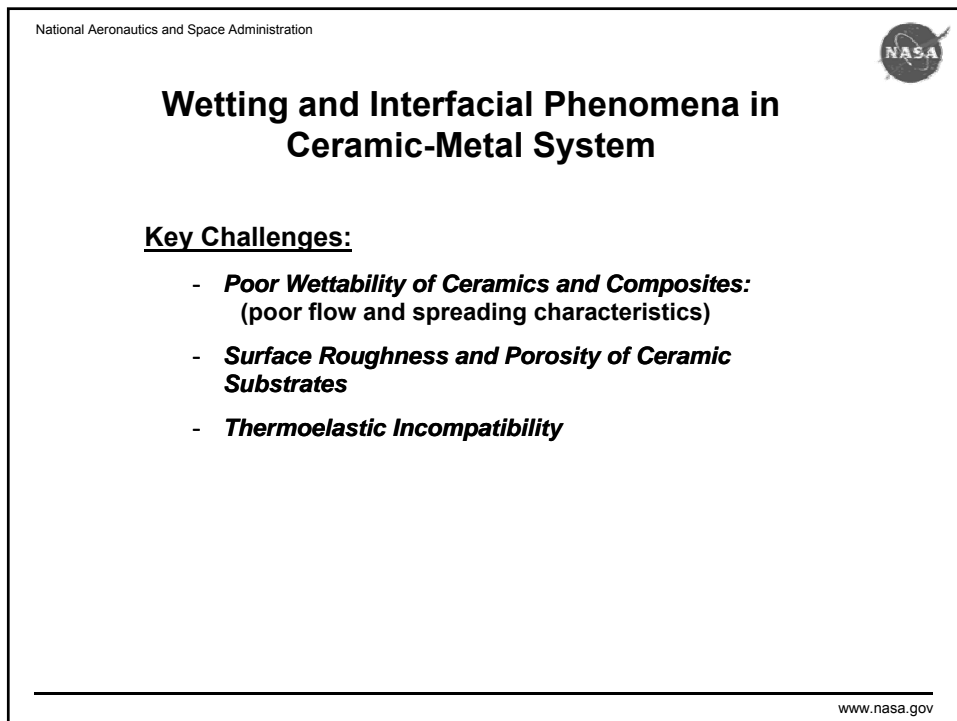
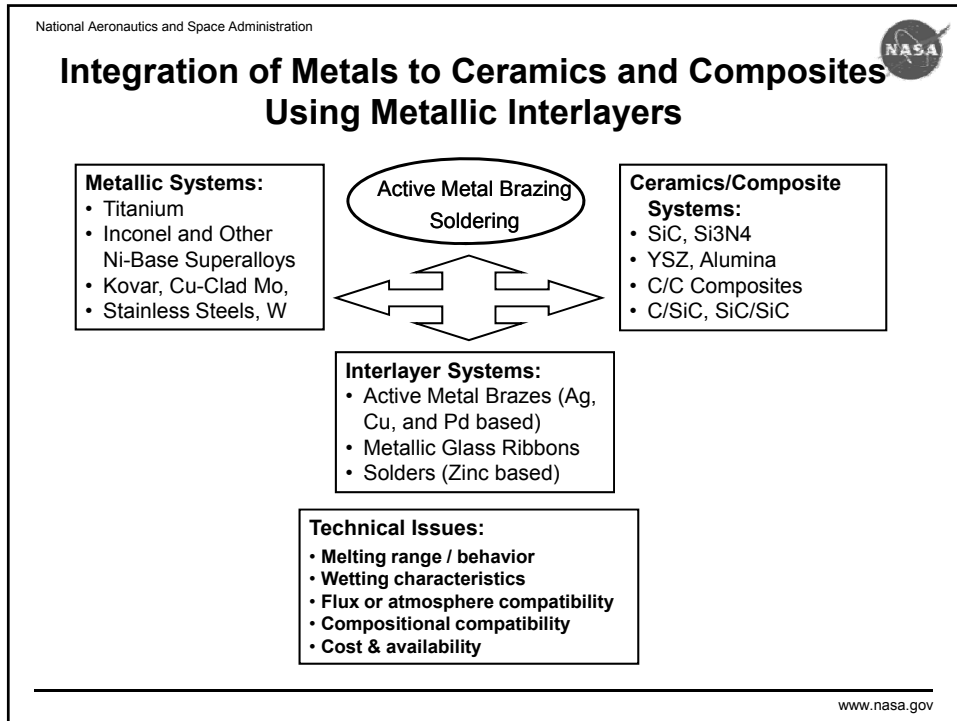
www.nasa.gov

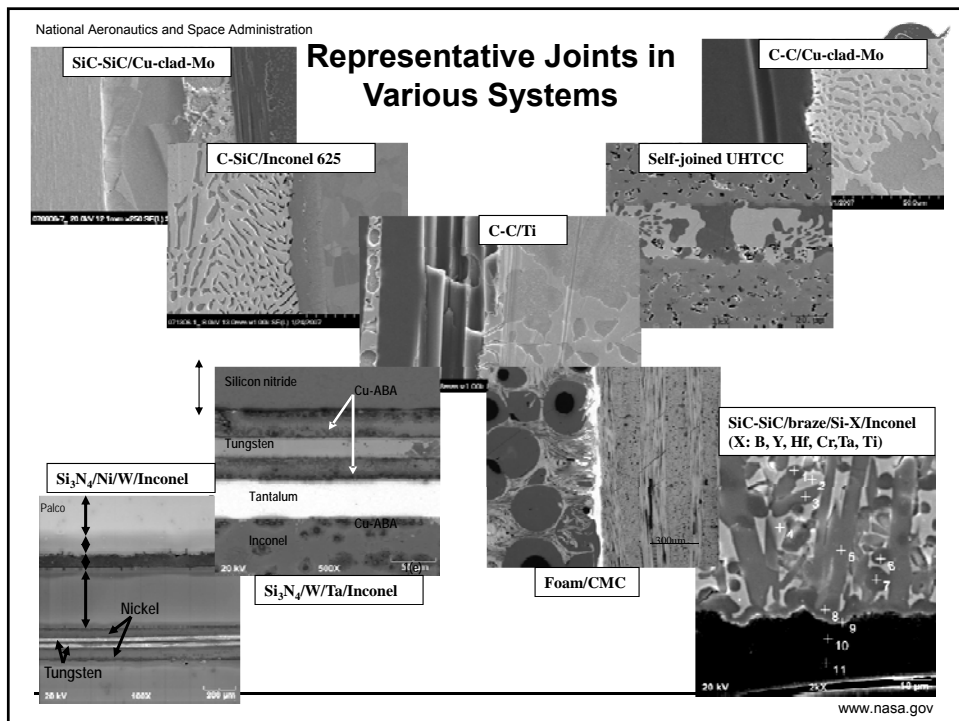
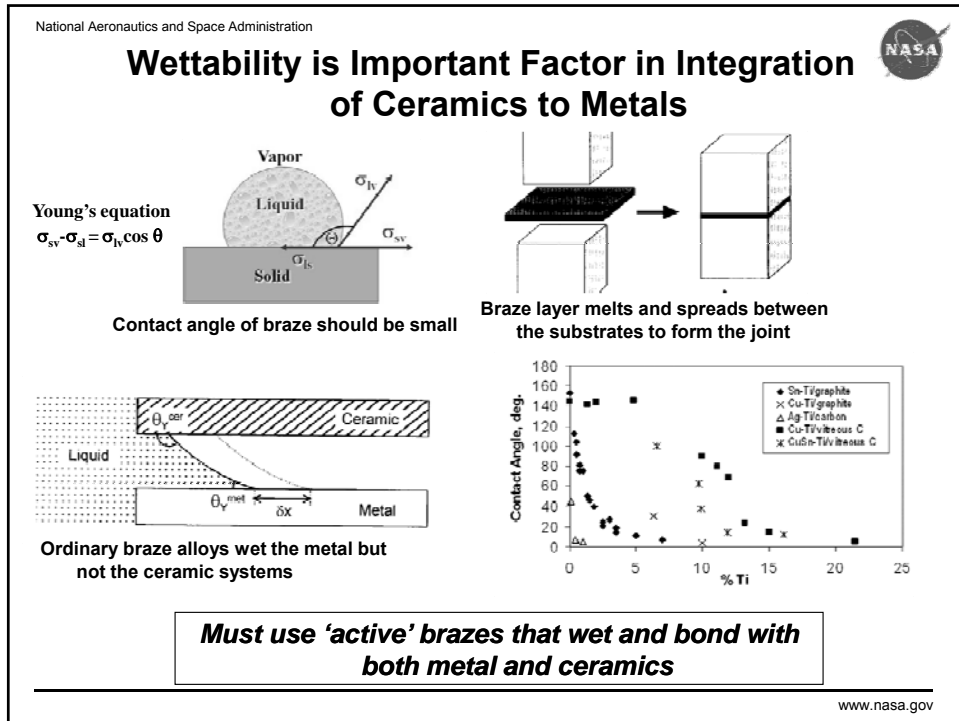












National Aeronautics and Space Administration



Integration Technologies for Improved Efficiency and Low Emissions

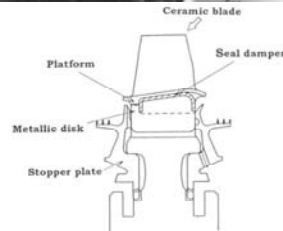
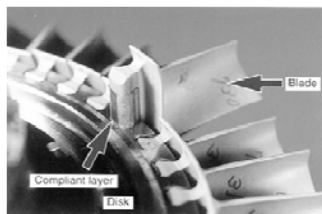
- Gas Turbine Components

www.nasa.gov

National Aeronautics and Space Administration

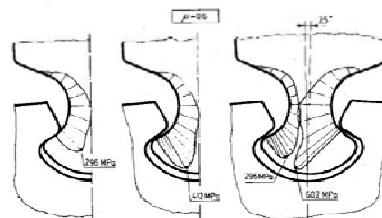


Advanced Silicon Nitride Based Components for Energy and Aerospace Systems



Hybrid Gas Turbine Blade
(Ceramic Blade and Metallic Disk)
in NEDO's Ceramic Gas Turbine
R&D Program, Japan (1988-1999)

Issues with Inserted Ceramic Blades



There are contact stresses at the metal-ceramic interface. Compliant layers (i.e. Ni-alloy+Pt) are used to mitigate the stress and damage. Failures can occur in the compliant layer.

Mark van Roode, Solar Turbines

www.nasa.gov

National Aeronautics and Space Administration



Integration Technologies for Silicon Nitride Ceramics to Metallic Components

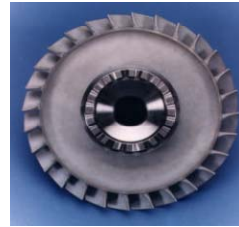
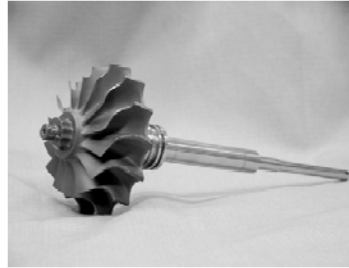
INTEGRAL ROTORS

- No Compliant Layer with Disk
- Attachment of Ceramic Rotor to Metal Shaft
- Primarily Small Parts
- Ability to Fabricate Larger Parts Has Improved
- Integral Rotors are Replacing Metal Disks with Inserted Blades



Mark van Roode, Solar Turbines

Industry Direction



IR Silicon Nitride Rotor, DOE Microturbine Program (top)
H-T. Lin, ORNL

www.nasa.gov

National Aeronautics and Space Administration



Integration of Silicon Nitride to Metallic Systems

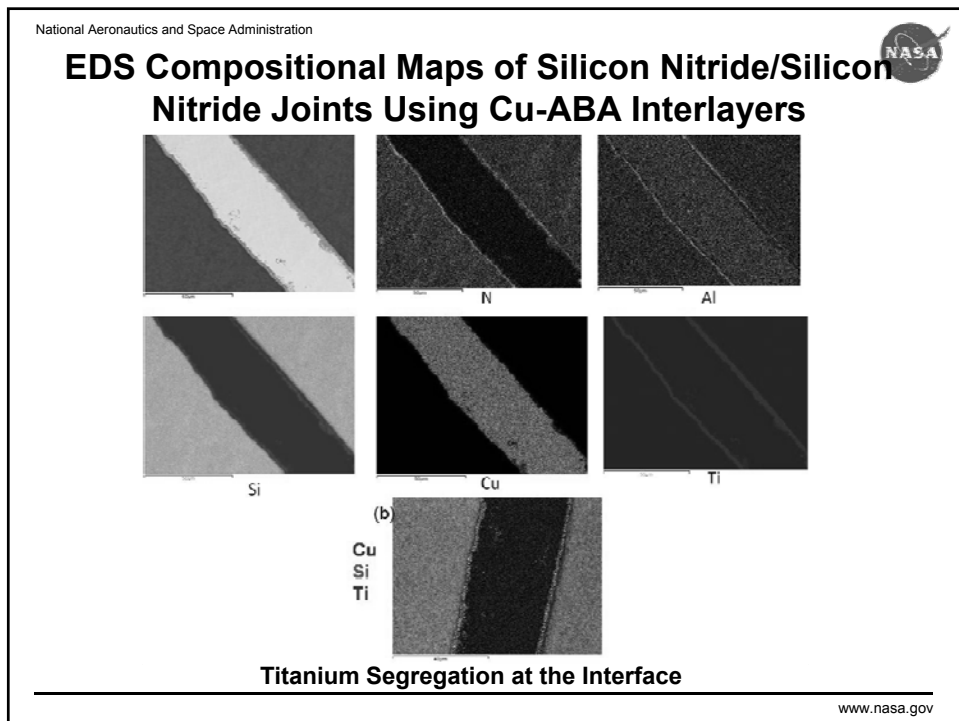
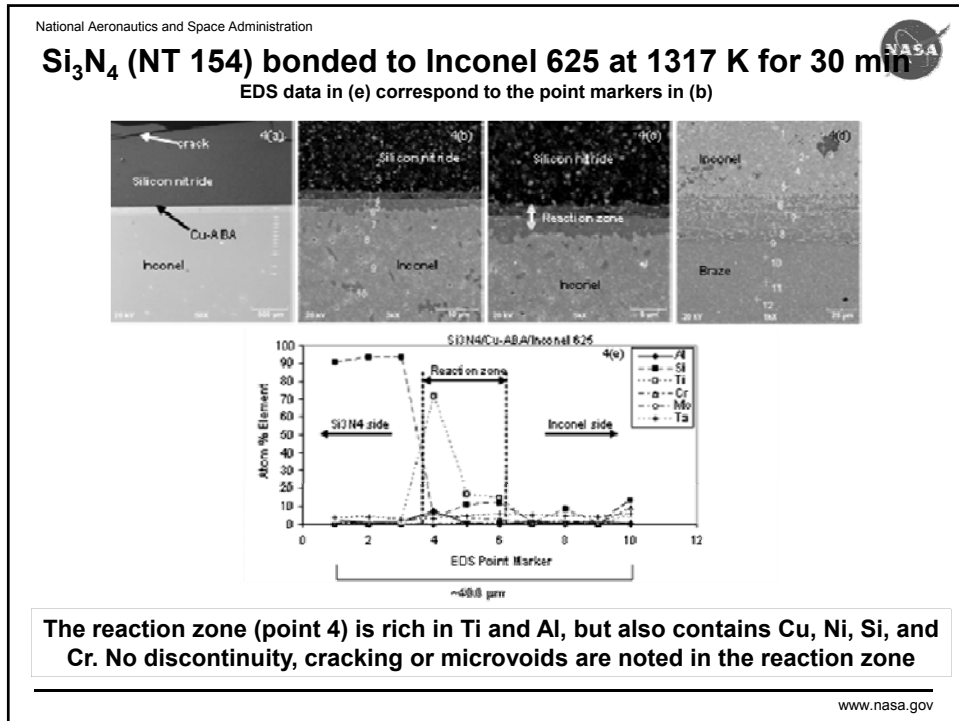
Approach: Use multilayers to reduce the strain energy more effectively than single layers.

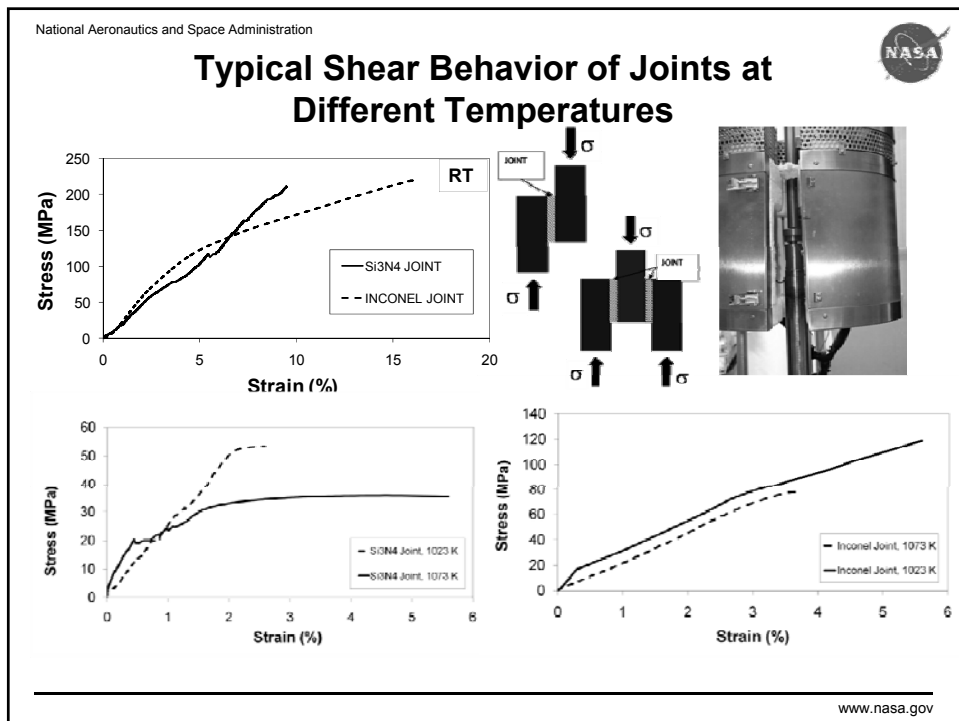
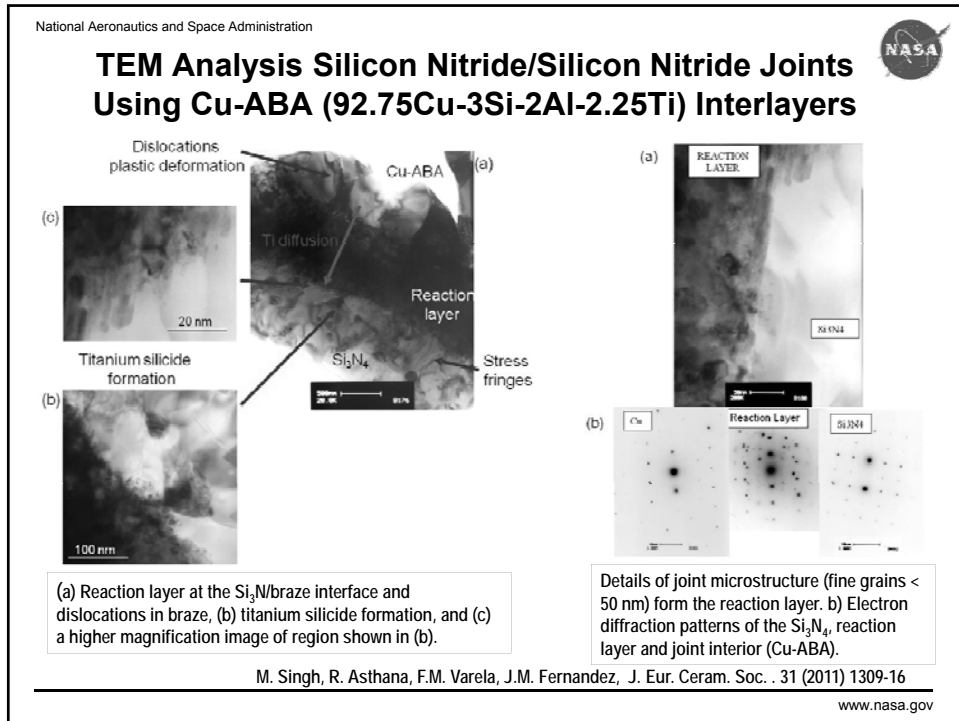
Challenge: Multiple interlayers increase the number of interfaces, thus increasing the probability of interfacial defects.

Material	CTE $\times 10^6/K$	Yield Strength, MPa
Silicon nitride	3.3	-
Inconel 625	13.1	-
Ta	6.5	170
Mo	4.8	500
Ni	13.4	14-35
Nb	7.1	105
Kovar	5.5-6.2	270
W	4.5	550

Various combinations of Ta, Mo, Ni, Nb, W and Kovar to integrate Silicon nitride to Nickel-Base Superalloys

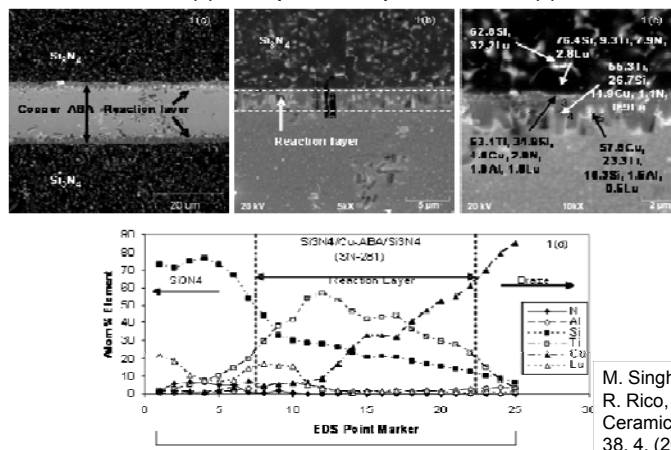
www.nasa.gov





**Kyocera SN-281 Si_3N_4 bonded at 1317 K for 5 min using
Cu-ABA (92.75Cu-3Si-2Al-2.25Ti) Interlayers**

EDS data in (d) correspond to the point markers in (b)



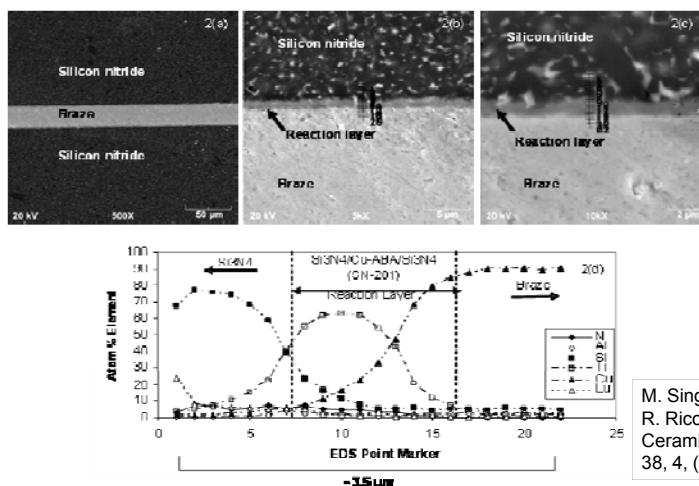
M. Singh, R. Asthana, J. R. Rico, J.M. Fernandez, Ceramic International, 38. 4. (2012) 2793-2802

- An inhomogeneous reaction layer (2-2.5 μm) comprising of a dark-gray Ti-Si phase, possibly titanium silicide, and a lighter Cu (Si, Ti) phase has developed.
- The product phase crystals are oriented perpendicular to the interface (growth direction).
- No interfacial excess of Lu: exists in minute quantities in reaction layer and increases toward Si_3N_4 .

www.nasa.gov

Kyocera SN-281 Si_3N_4 bonded at 1317 K for 30 min

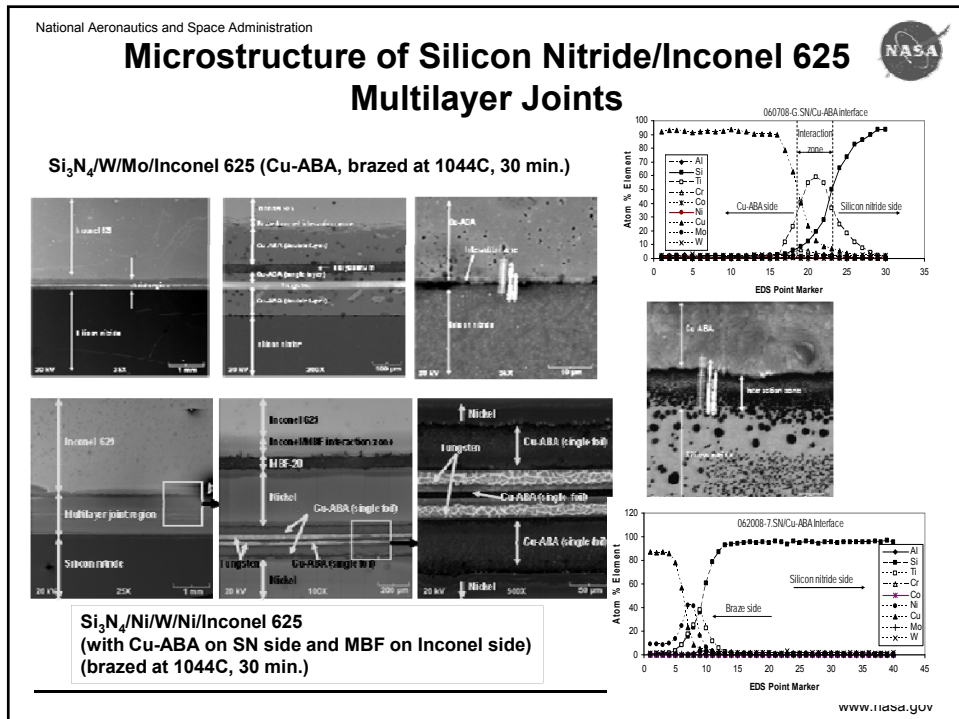
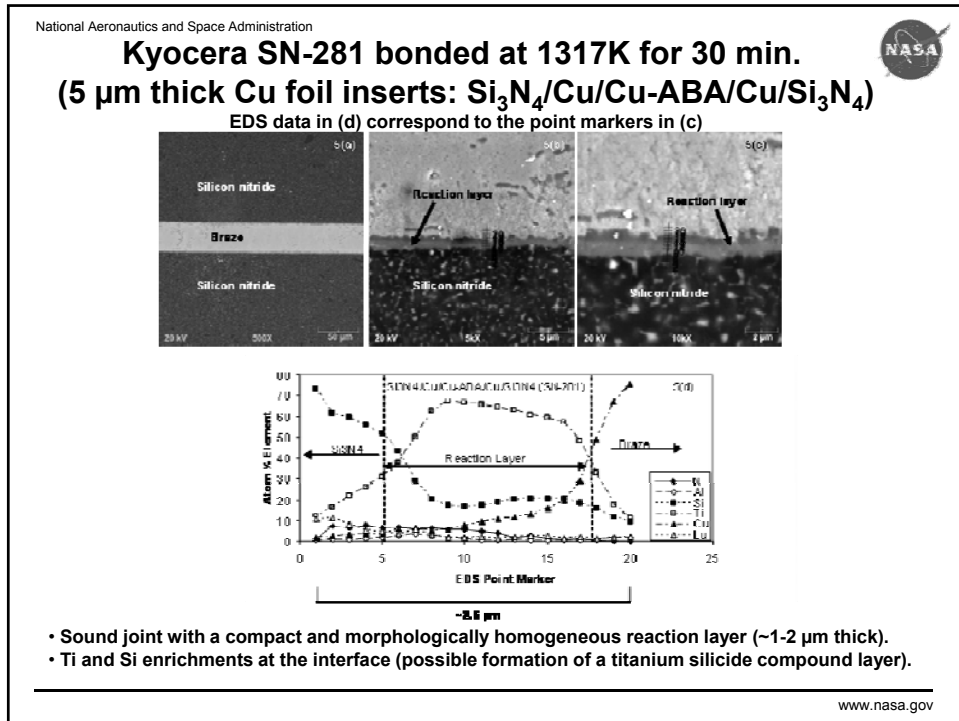
EDS data in (d) correspond to the point markers in (C)

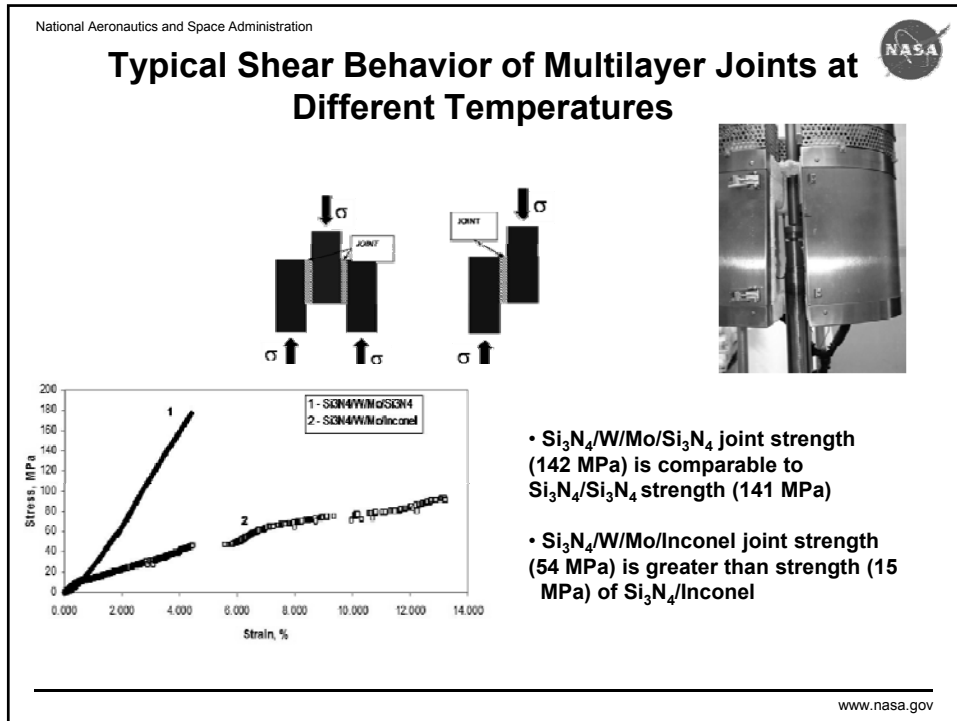


M. Singh, R. Asthana, J. R. Rico, J.M. Fernandez, Ceramic International, 38, 4, (2012) 2793-2802

- No increase in reaction layer thickness for 30 min. (faster kinetics in the early stages of reaction).
- Morphologically a more homogeneous, compact, and featureless reaction layer (possible coalescence of coarsened silicide crystals).

www.nasa.gov





National Aeronautics and Space Administration

NASA

Integration Technologies for Improved Efficiency and Low Emissions

- MEMS-LDI Fuel Injector

www.nasa.gov

National Aeronautics and Space Administration



Integration Technologies for MEMS-LDI Fuel Injector

Objective: Develop Technology for a SiC Smart Integrated Multi-Point Lean Direct Injector (SiC SIMP-LDI)

- Operability at all engine operating conditions
- Reduce NOx emissions by 90% over 1996 ICAO standard
- Allow for integration of high frequency fuel actuators and sensors

Possible Injector Approaches

1. Lean Pre-Mixed Pre-Evaporated (LPP)

Advantages - Produces the most uniform temperature distribution and lowest possible NOx emissions

Disadvantages - Cannot be used in high pressure ratio aircraft due to auto-ignition and flashback

2. Lean Direct Injector (LDI)

Advantages - Does not have the problems of LPP (auto-ignition and flashback)

- Provides extremely rapid mixing of the fuel and air before combustion occurs

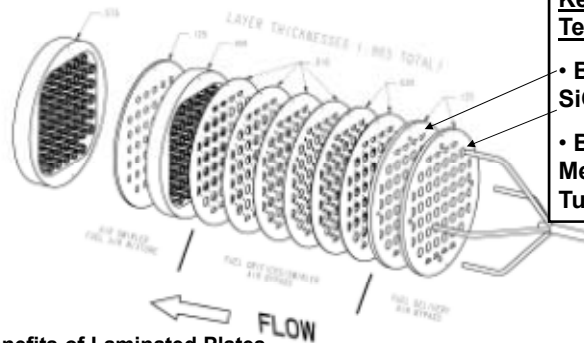
www.nasa.gov

National Aeronautics and Space Administration



Lean Direct Injector Fabricated by Bonding of SiC Laminates

SiC laminates can be used to create intricate and interlaced passages to speed up fuel-air mixing to allow lean-burning, ultra-low emissions



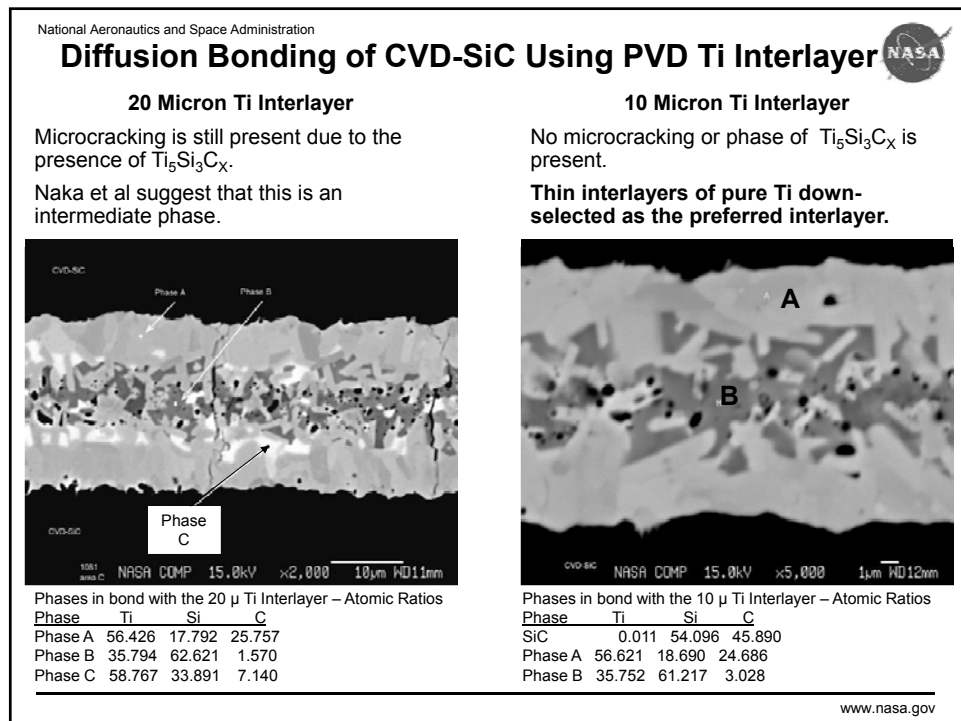
Key Enabling Technologies:

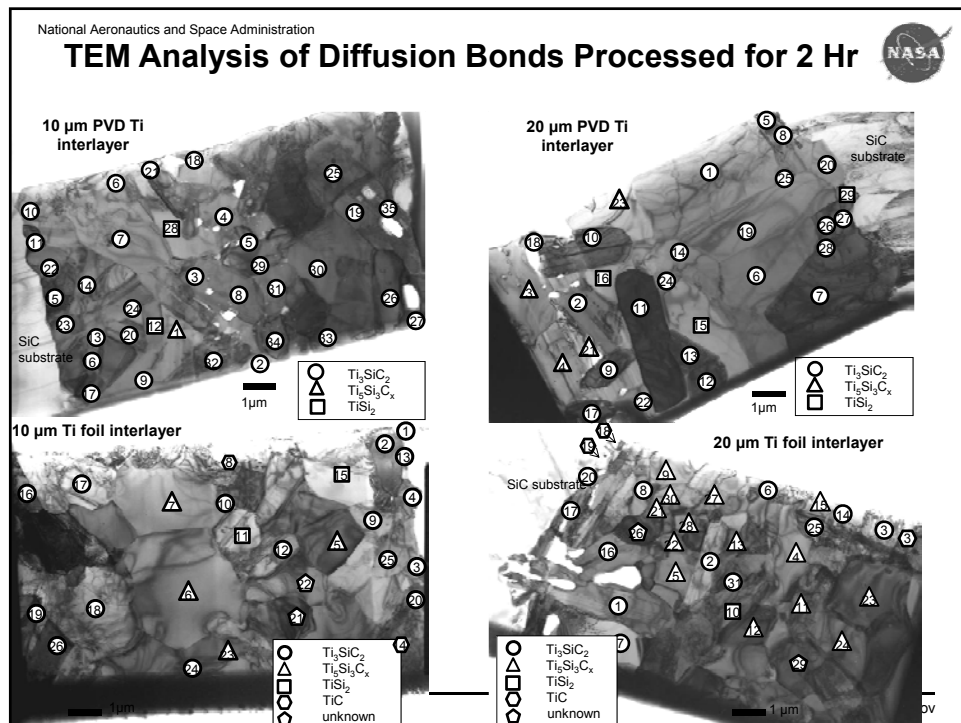
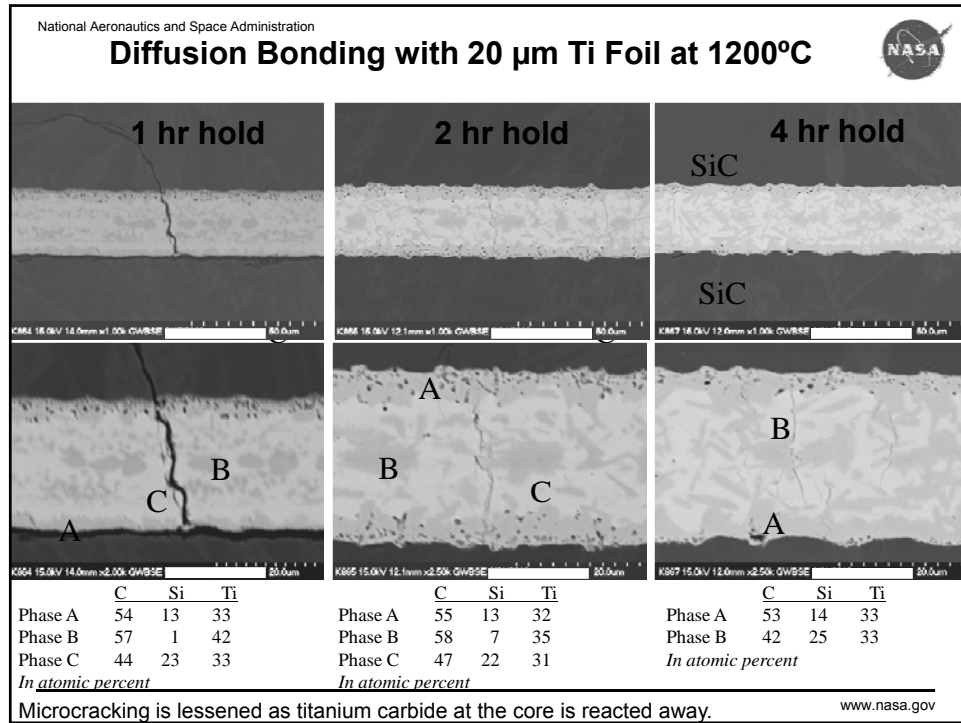
- Bonding of SiC to SiC
- Brazing of SiC to Metallic (Kovar) Fuel Tubes

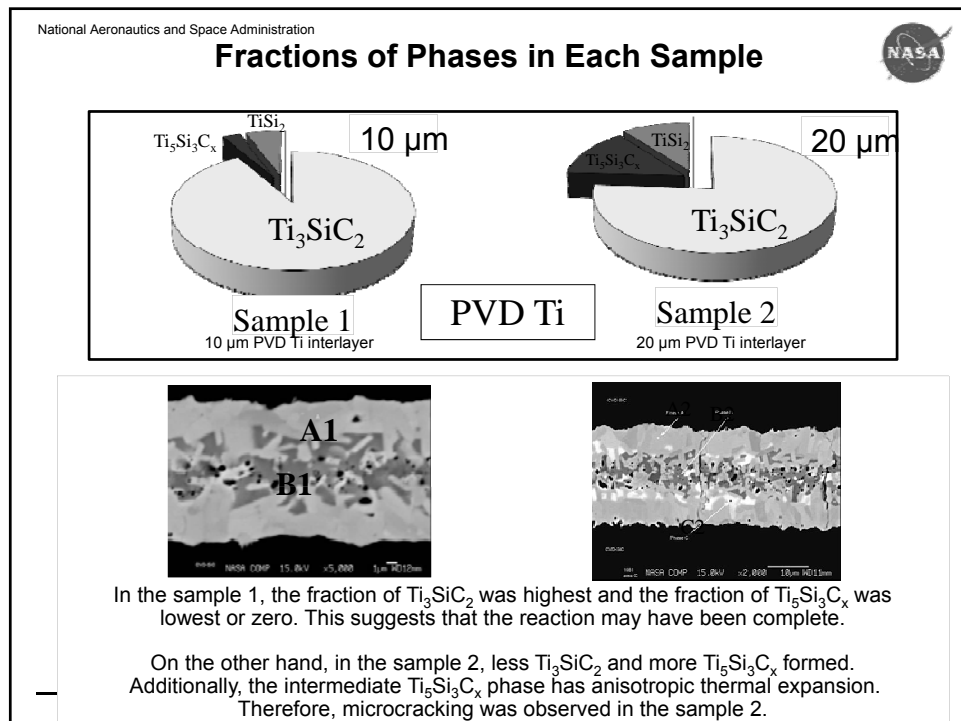
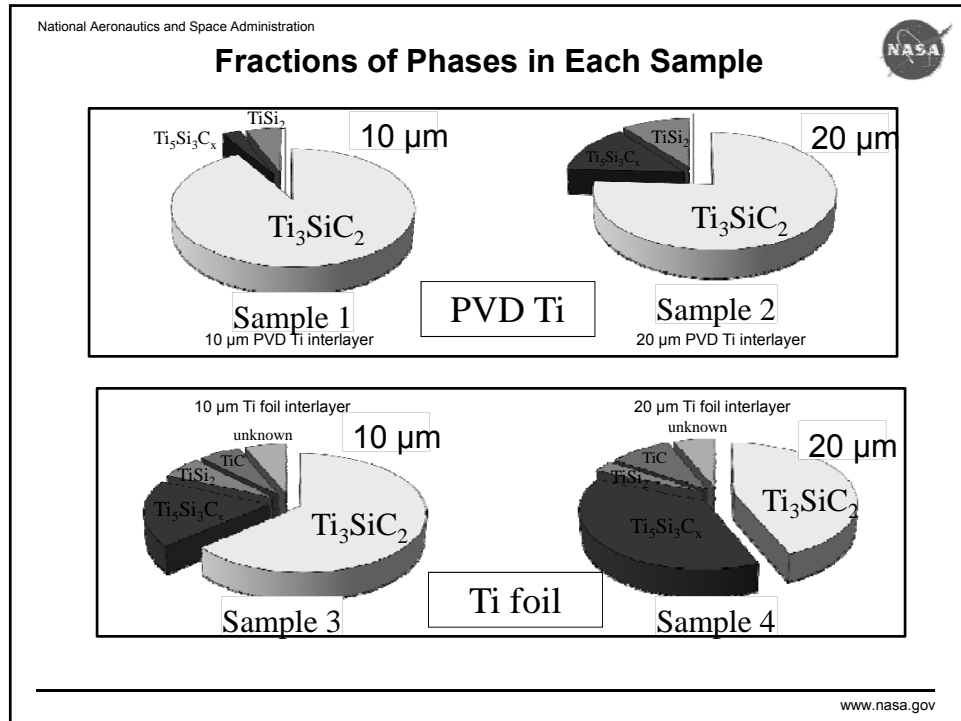
Benefits of Laminated Plates

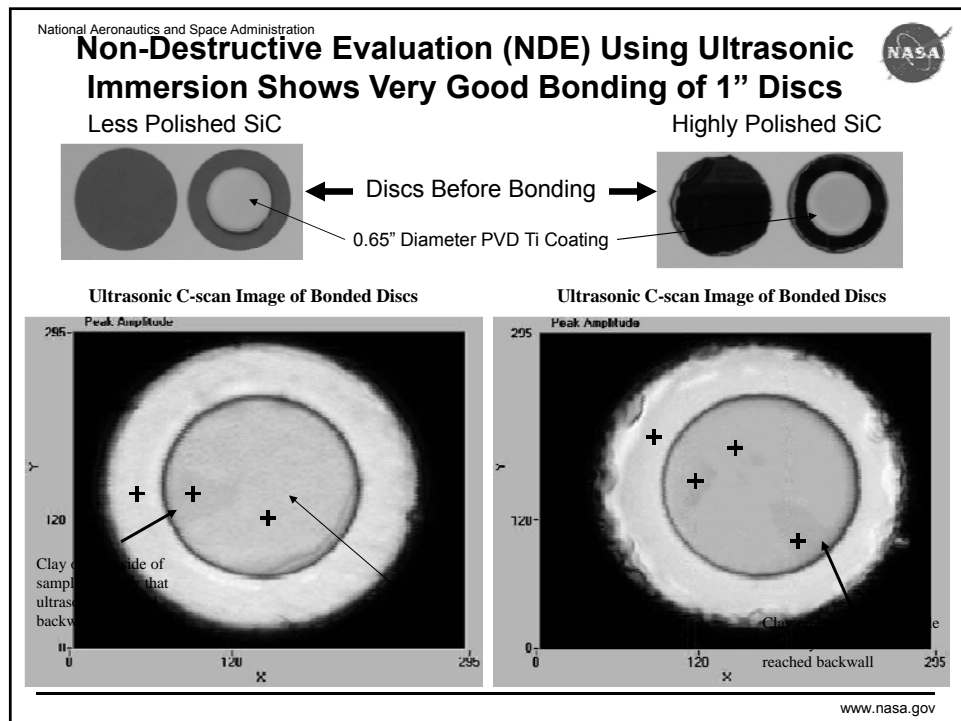
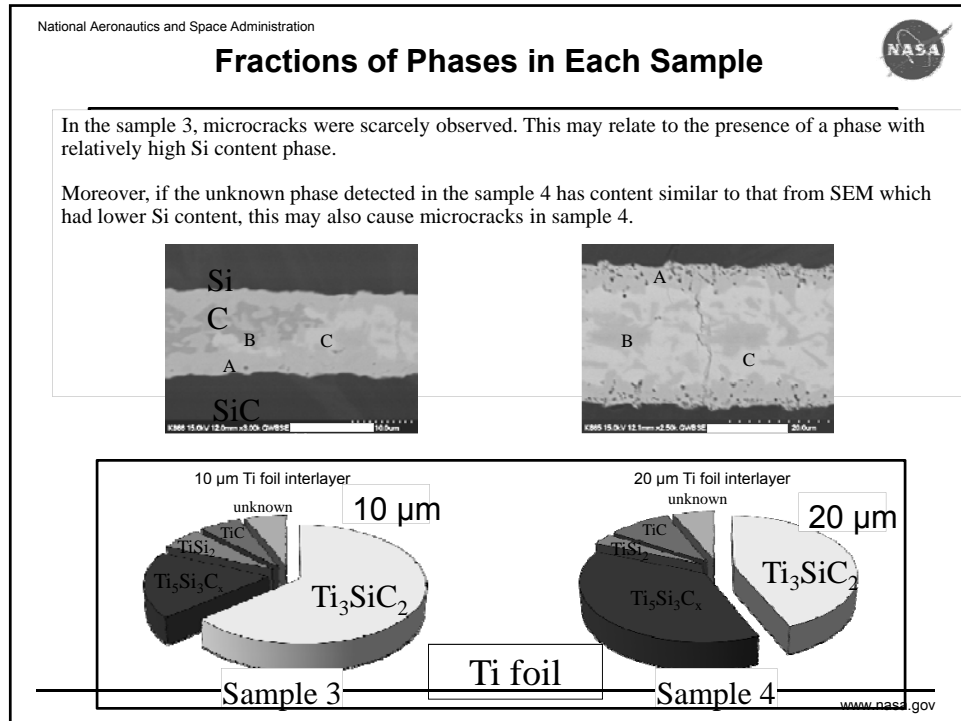
- Passages of any shape can be created to allow for multiple fuel circuits
- Provides thermal protection of the fuel to prevent choking
- Low cost fabrication of modules with complicated internal geometries through chemical etching

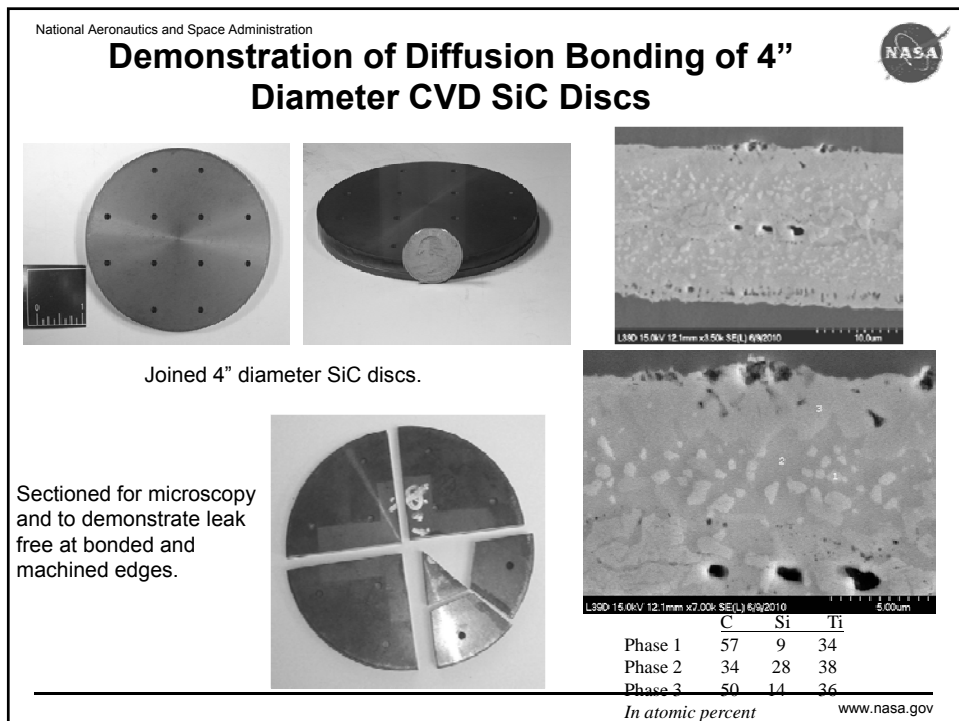
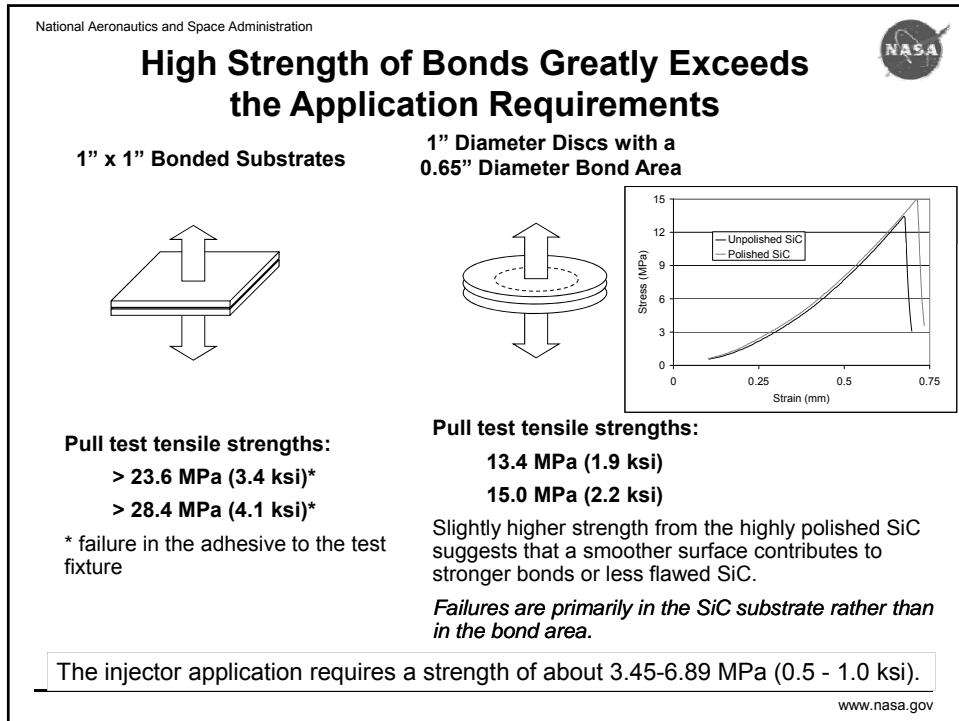
www.nasa.gov

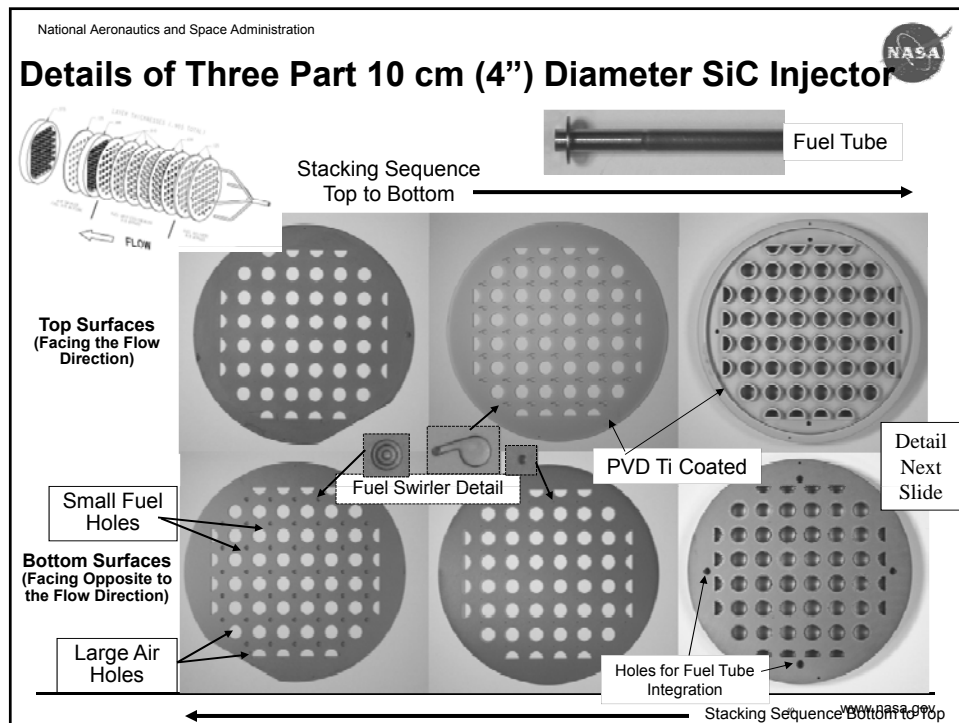
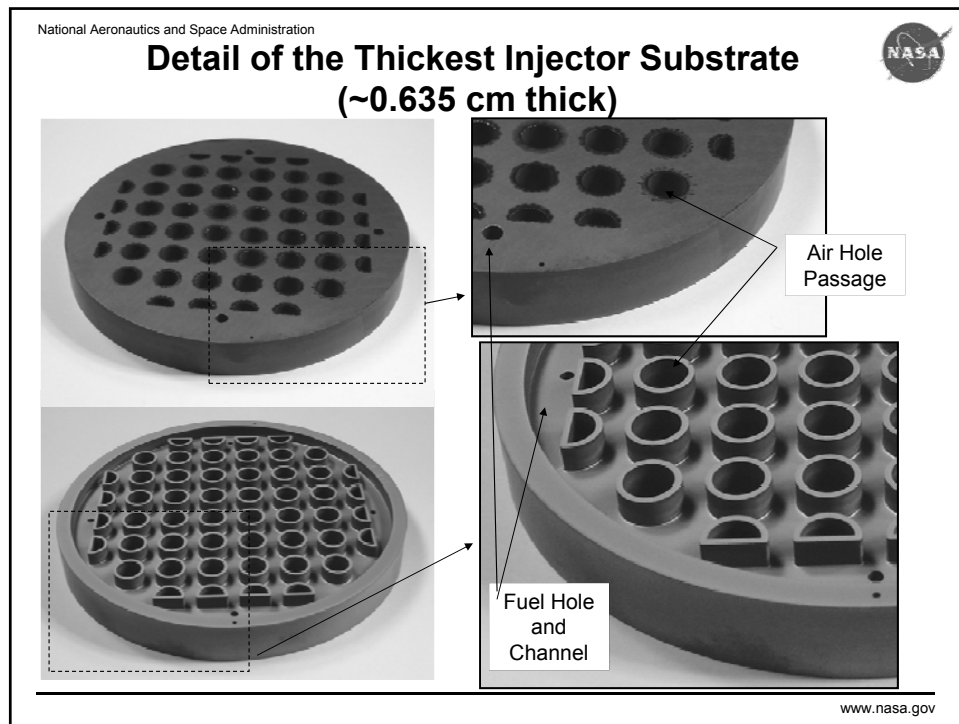


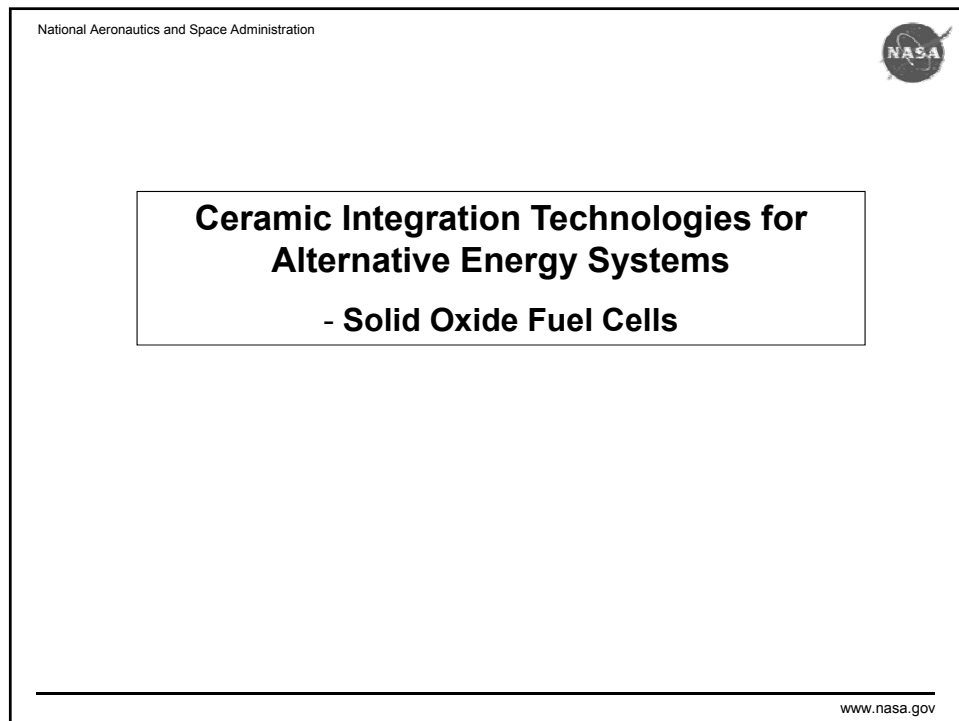
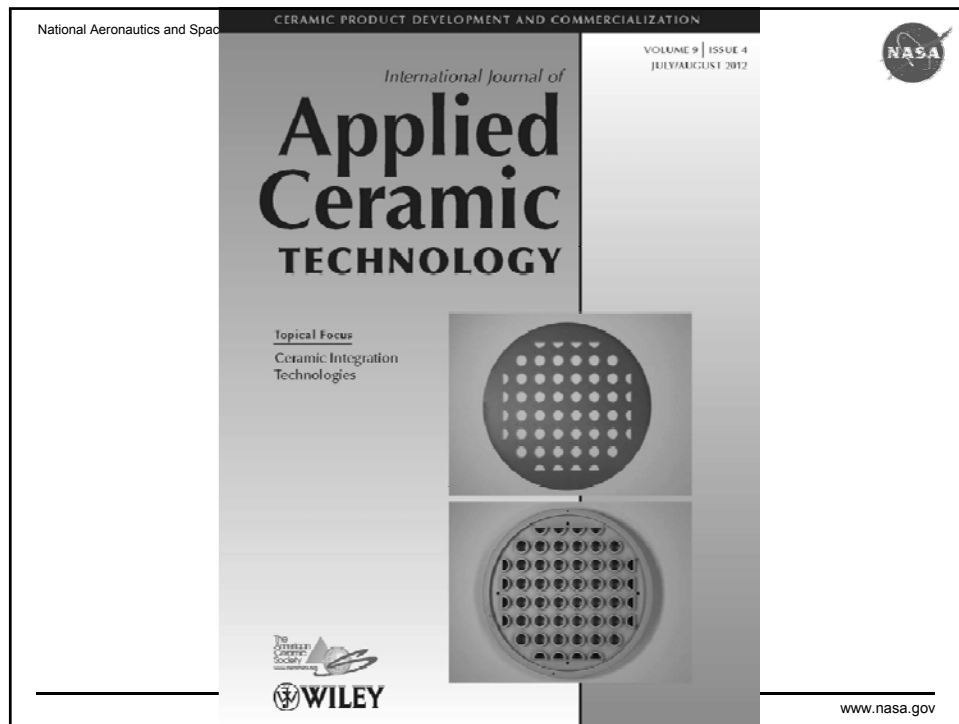


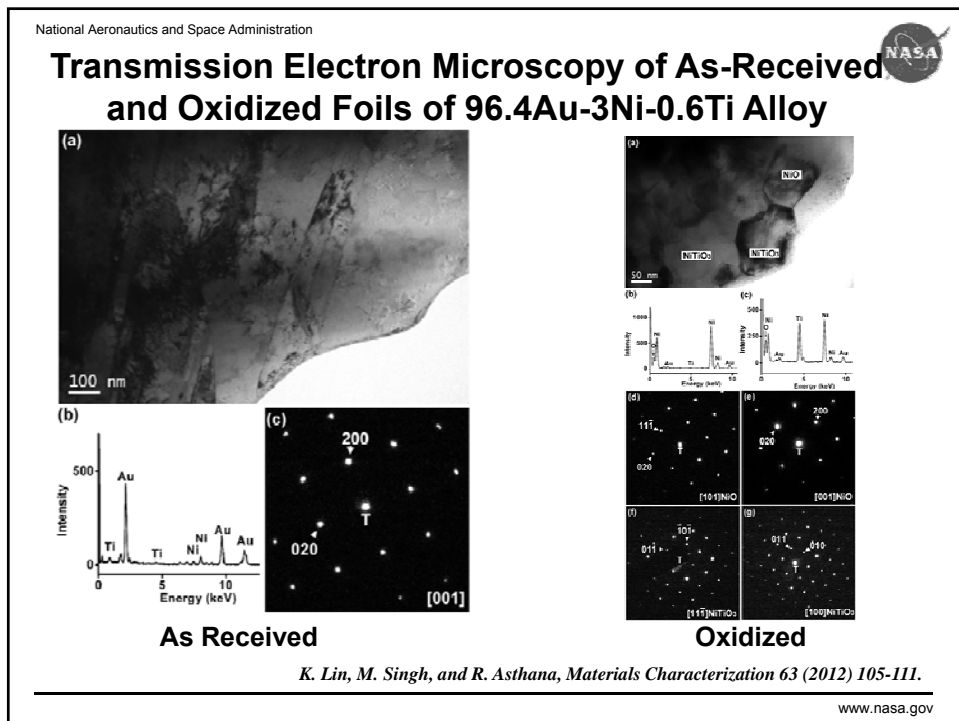
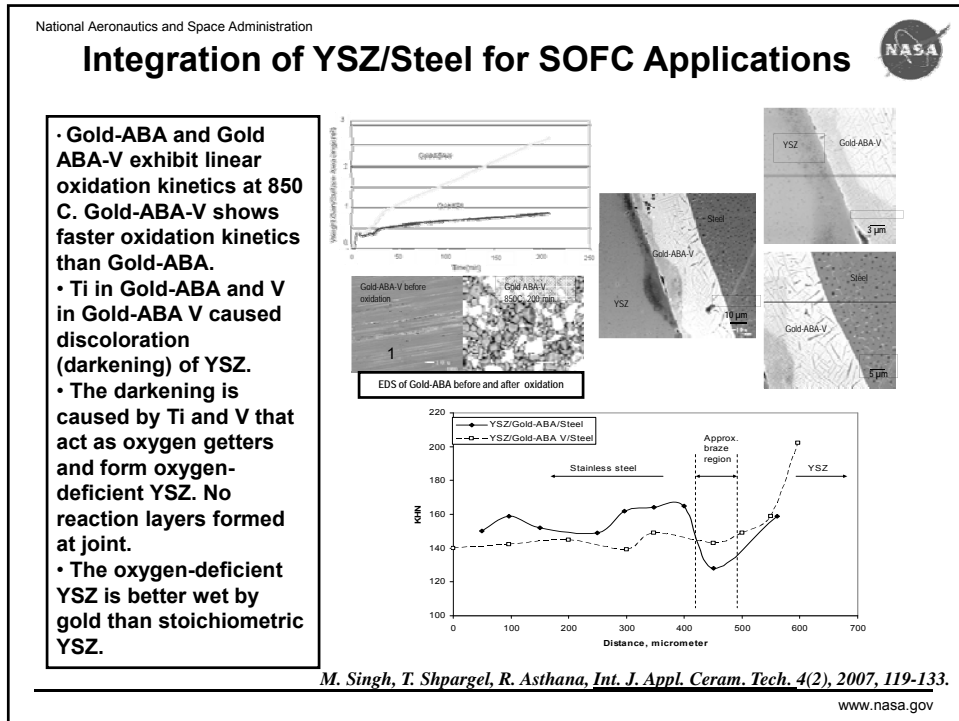


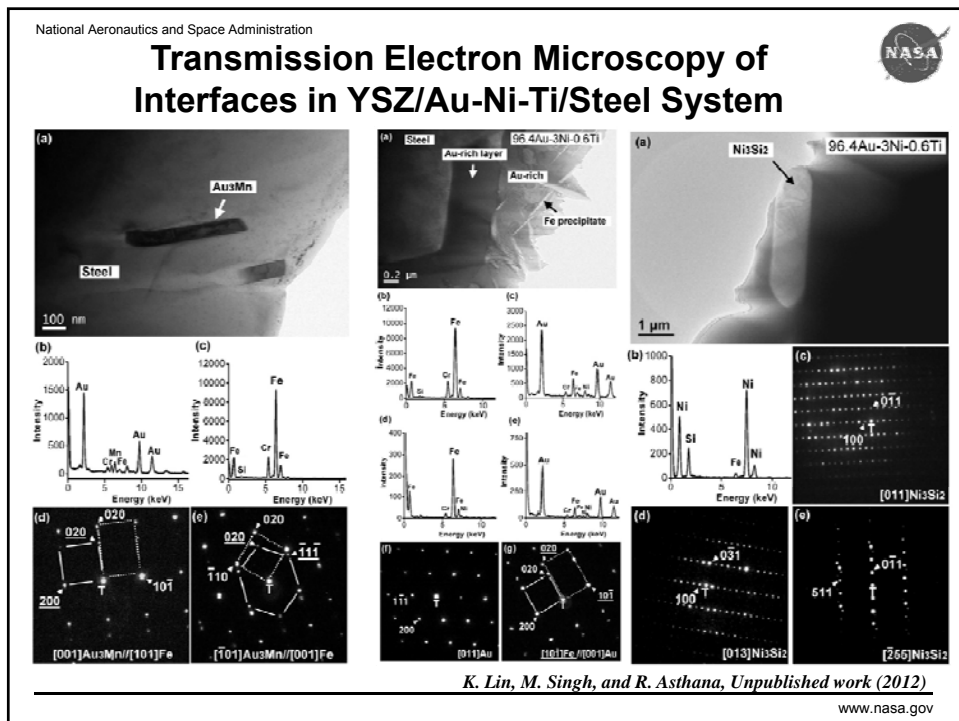
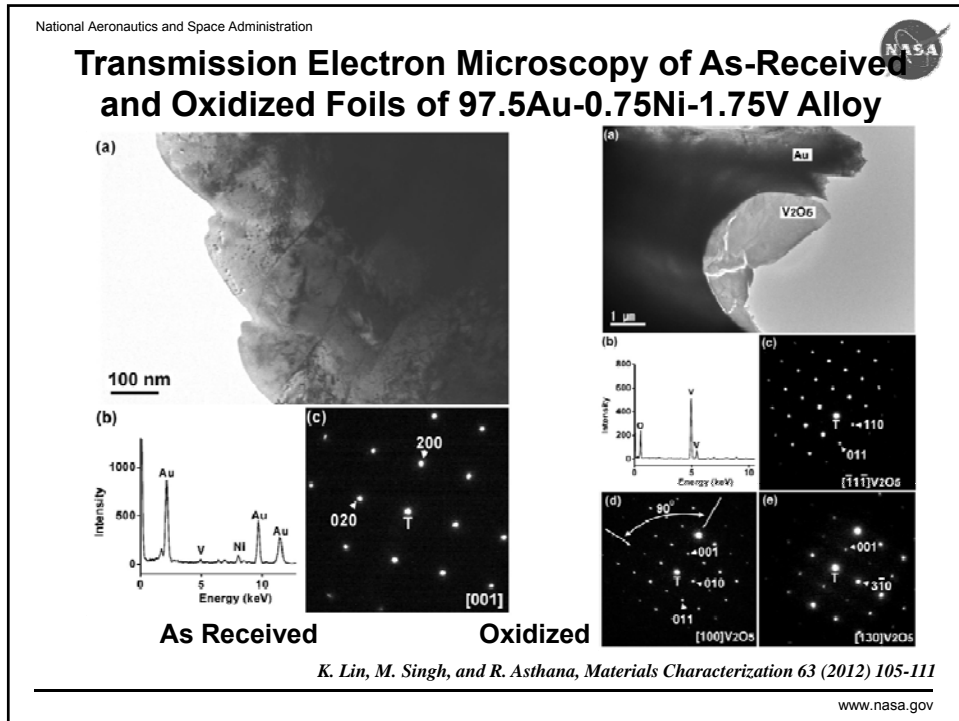












National Aeronautics and Space Administration



Integration of YSZ/Steel for SOFC Applications

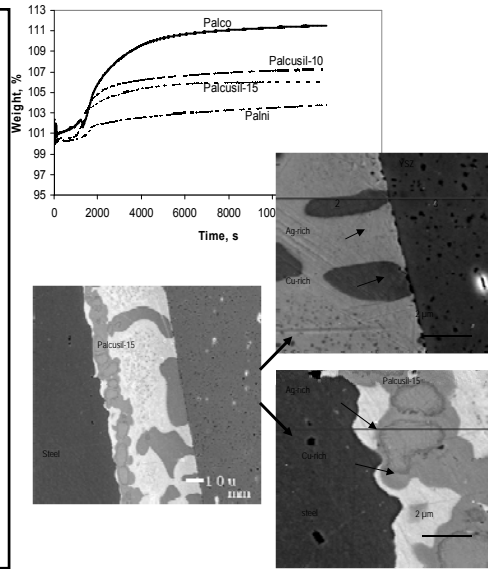
• Pd-base brazes (Palco and Palni) and Ag-base brazes (Palcusil-10 and Palcusil-15) were characterized for oxidation at 750°C.

• Structural changes accompany oxidation which is fastest for Palco, slowest for Palni, and intermediate for Palcusil-10 and Palcusil-15.

• All brazes were effective in joining yttria stabilized zirconia (YSZ) to stainless steel for solid oxide fuel cell (SOFC).

• Dissolution of YSZ and steel in braze, and braze constituents in YSZ and steel led to diffusion and metallurgically sound joints.

• Knoop hardness (HK) profiles are similar for all brazes, and exhibit a sharp discontinuity at the YSZ/braze interface.



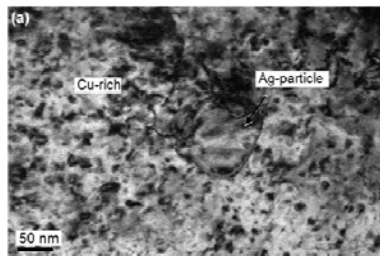
M. Singh, T. Shpargel, R. Asthana, Mater. Sci. Eng. A 485 (2008) 695-702

www.nasa.gov

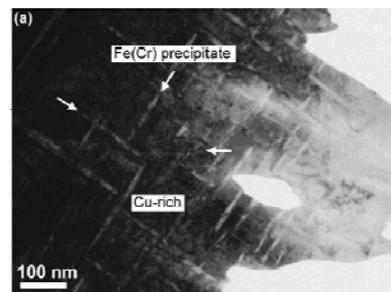
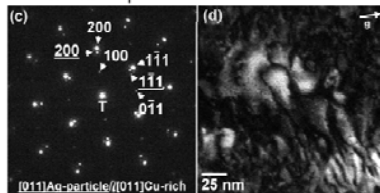
National Aeronautics and Space Administration



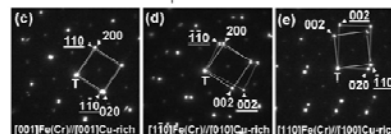
Transmission Electron Microscopy of Interfaces in YSZ/Ag-Cu-Pd/Steel System



(b) Atomic %	Ag	Cu	Fe	Cr	Pd	Zr
Ag-particle	72.8	12.5	7.3	2.5	1.3	3.7
Cu-rich	1.8	70.9	6.3	1.0	18.0	2.0

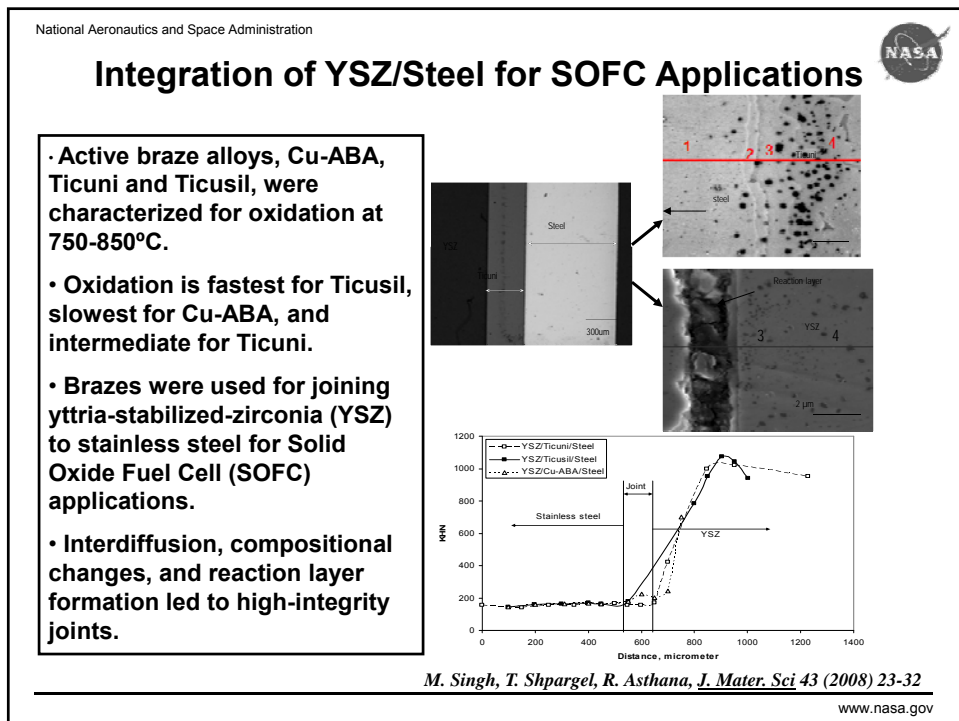
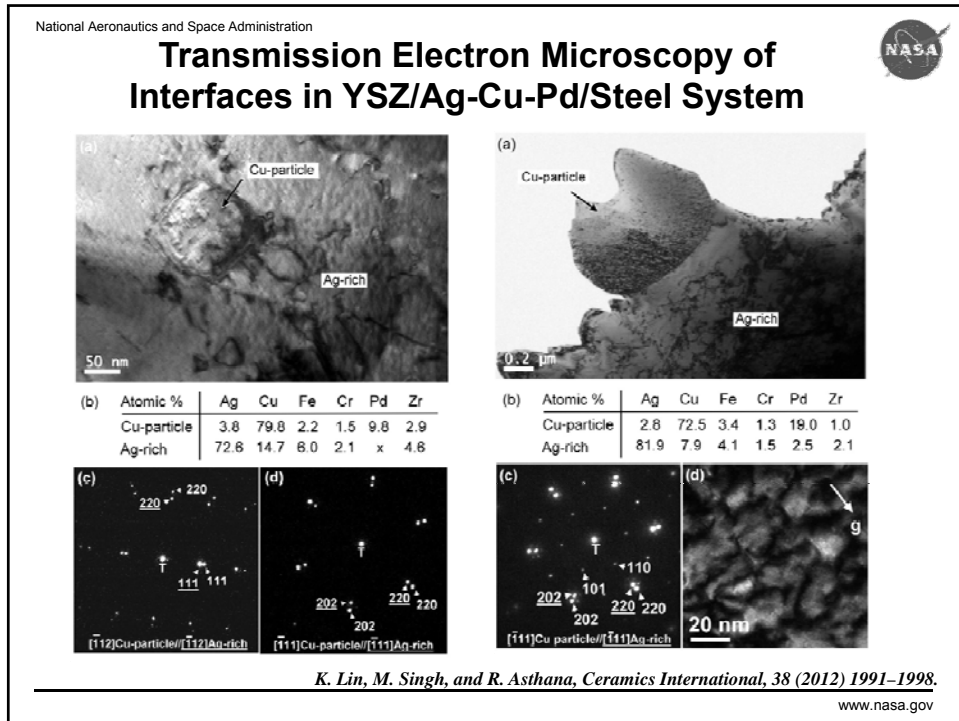


(b) Atomic %	Ag	Cu	Fe	Cr	Pd	Zr
Fe(Cr) precipitate	0.3	3.1	75.7	16.3	4.4	0.2
Cu-rich	3.5	31.2	24.8	4.4	35.0	1.1



K. Lin, M. Singh, and R. Asthana, Ceramics International, 38 (2012) 1991-1998.

www.nasa.gov





Concluding Remarks

- **Ceramic integration technologies are critically needed for the successful development and applications of ceramic components in a wide variety of energy applications.**
- **Major efforts are needed in developing joint design methodologies, understanding the size effects, and thermomechanical performance of integrated systems in service environments.**
- **Development of life prediction models for integrated components is needed for their successful implementation. In addition, global efforts on standardization of integrated ceramic testing and standard test method development are also required.**



Acknowledgements

- **Prof. Rajiv Asthana, University of Wisconsin-Stout**
- **Mr. Craig E. Smith, Ohio Aerospace Institute**
- **Mr. Michael H. Halbig, NASA Glenn Research Center**
- **Prof. J.M. Fernandez, University of Seville, Spain**
- **Mr. Ron Phillips, ASRC Corp.**
- **Mr. Ray Babuder, Case Western Reserve University**
- **A number of summer students**